

## ***Fiber Reinforced Polymer Composites***

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MSE 383 Unit 3-5  
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### ***Scope***

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#### **General Introduction**

composite constituents  
applications

#### **Strength and Fracture: Analysis**

#### **Composite design**

#### **Fabrication of Composites (Unit 4)**

### ***Learning Objectives***

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How properties of  $E$ ,  $\sigma$  and fracture toughness of FRP's are explained and predicted  
properties of fibers and matrix  
fiber-matrix interface  
pattern of orientation of the fibrous phase

### ***Definition of a Composite***

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The combination of different, relatively homogenous materials to produce a heterogeneous materials to produce a heterogeneous material of a more complex structure displaying properties which none of its constituents can exhibit in isolation.

### ***Composite Constituents***

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Matrix materials (thermoplastics & thermosetting resins)  
Fibers (particulate or fibrous glass or carbon etc.)  
Fiber-Matrix interface  
Fiber orientation  
Voids

## *Polymer Composites vs. Conventional Materials*

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Lightweight  
High specific stiffness and strength  
High toughness  
Corrosion resistant  
High dimensional stability  
Ease of fabrication  
Tailorable mechanical and thermal properties

## *Applications of Composites*

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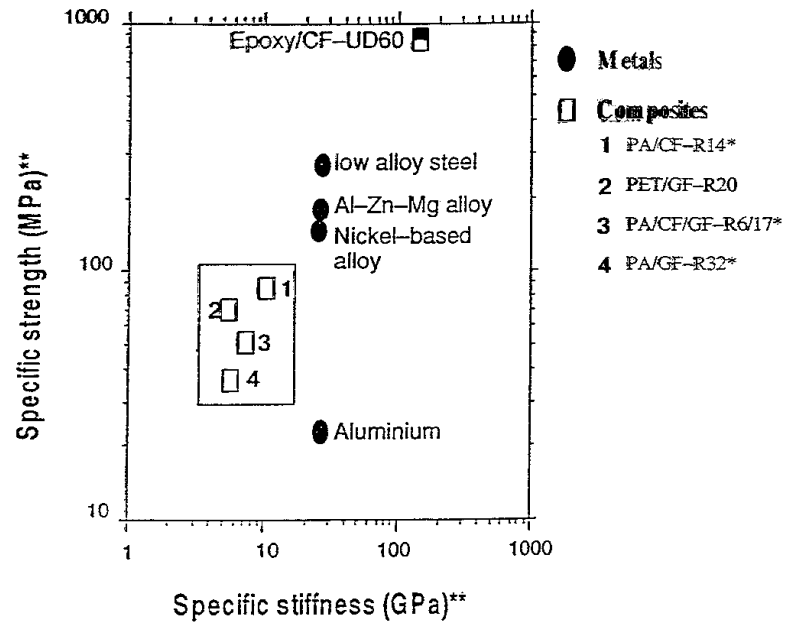
**Aerospace** (Carbon, Kevlar<sup>®</sup>, GF composites in nose cones, helicopter blades, body panels, etc.)  
**Automotive Engineering** (GFRP in Pontiac Fiero, for instance, in front-end moldings, fascias, bumper back-up beams, etc.)  
**Bioengineering** (CFRP in prosthetic applications such as orthopedic fracture fixation plates, Mandibular prosthesis, e.g., jaw remodeling)  
**Chemical Engineering** (GFRP in chemical plant for pressure vessels, valves, etc.)  
**Structural/Civil Engineering** (GRC in the building industry etc.)  
**Domestic** (injection molded reinforced thermoplastics & polyester molding compounds in furniture, television, & computer casings, kitchen equipment, crash helmets, etc.)  
**Electrical Engineering** (GFRP in high strength insulators, structural components for switch gear, etc.)  
**Marine Engineering** (GFRP in pleasure craft, etc.)  
**Sport** (CFRP & boron fiber composites in vaulting poles, tennis rackets, golf clubs, surf & skate boards, etc.)  
ILLUSTRATE

## *Fiber Characteristics*

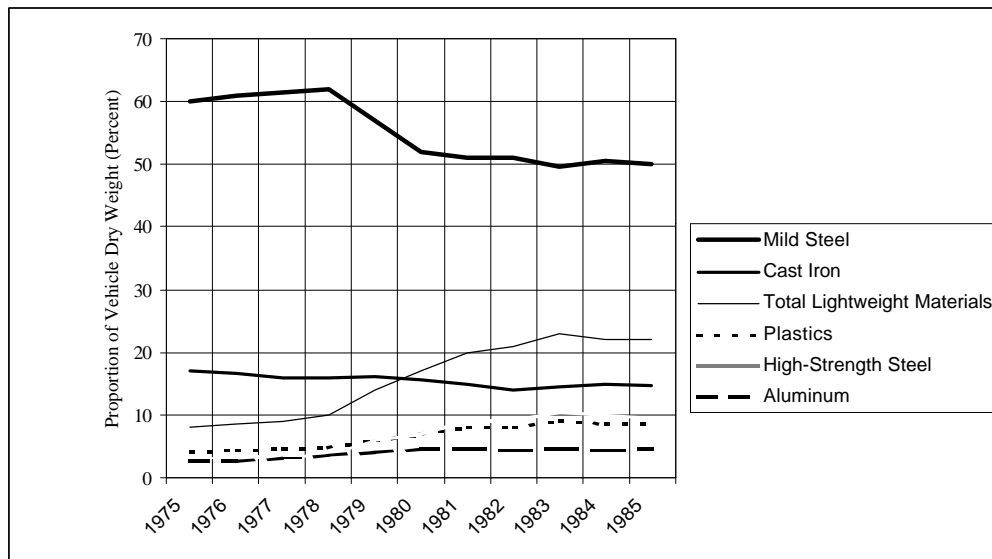
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Inorganic Fibers and Whiskers (e.g., metal filaments and Al<sub>2</sub>O<sub>3</sub> whiskers are used to reinforce metals)  
High Modulus Organic Fibers (e.g., poly paraphenylene terephthalate - KEVLAR - are used in polymers)  
SHOW TABLE

## Fiber Characteristics, Cont'd

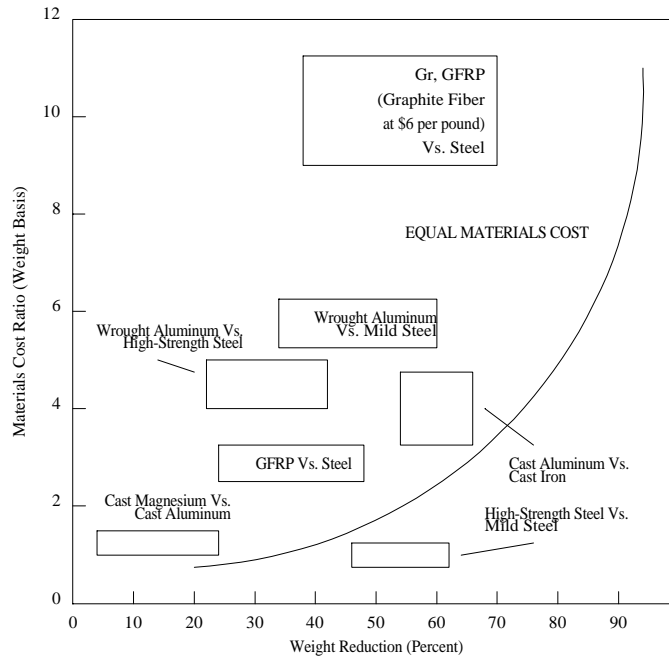


Comparison of specific stiffness and strength of some engineering materials  
 (\*this work, other data taken from Hull (1981) and Piggott (1980); \*\*unit of density is Mg/m<sup>3</sup>)



LIGHTWEIGHT MATERIALS CONTENT of the average U.S. vehicle has increased sharply since the mid-1970's. The data are for a sales-weighted average of the Ford U.S. passenger fleet. Plastics, aluminum and high-strength steel currently account for about a fourth of the dry weight of a Ford passenger car. These lightweight materials began supplanting cast iron and mild steel in U.S. vehicles about a decade ago. Manufacturers were then faced with the weight increases that resulted from technology adopted to meet Federal safety and emissions regulations and with rising fuel costs. In response they decreased the size of their vehicles and also increased the lightweight materials content.

## Fiber Characteristics, Cont'd



**COST OF SUBSTITUTING** a lightweight material depends on the weight saving and the prices of the two materials. The horizontal axis shows the weight reduction made possible by each substitution and the vertical axis shows the cost of the lightweight material in relation to its conventional counterpart. (GFRP is plastic reinforced with glass fibers. Gr,GFRP is plastic reinforced with fibers of graphite and glass; the data for that material assume a cost for graphite fiber of \$6 per pound, a third of its current cost.) For certain substitutions the two materials will be equal in cost. Most lightweight materials have not yet reached the break-even point; an exception is high-strength steel. In many instances, however, savings in processing and assembly costs for the lightweight material offset the additional cost of the material. The data were compiled by Magee.

*Fiber Characteristics, Cont'd*

**Weight Savings (%)**

High-Strength Steel		0%	Equal Volume	
Aluminum		65%		
Fiber-Reinforced Plastic	30 Kpsi Strength	81%		
Fiber-Reinforced Plastic	100 Kpsi Strength	81%		
Mild Steel		0%	Equal Collapse Load and Bending Stiffness	
High-Strength Steel		18%		
Aluminum		50%		
Fiber-Reinforced Plastic		30 Kpsi Strength		25%
Fiber-Reinforced Plastic	100 Kpsi Strength	60%		
Mild Steel		0%	Equal Stiffness	
High-Strength Steel		0%		
Aluminum		52%		
Fiber-Reinforced Plastic		30 Kpsi Strength		38%
Fiber-Reinforced Plastic	100 Kpsi Strength	48%		

Hypothetical aluminum part, the weight saving would be 56 percent. In neutral design situations the weight saving offered by the substitution of aluminum for cast iron ranges from 35 to 60 percent. Similarly, aluminum and fiber-reinforced plastics are much lighter than mild (ordinary) steel by volume. The weight savings, however, are much smaller if equal stiffness or equal collapse load and bending stiffness (a measure of structural strength) is needed. High-strength steel is no lighter by volume than mild steel, nor is it stiffer. Where structural strength is the main concern, however, high-strength steel does offer a weight saving: 18 percent in this example. The hypothetical examples in the illustration were developed by Christopher L. Magee of the Ford Motor Company.

## *Fiber Characteristics, Cont'd*

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Table 1: Lightweight Materials Weight Savings (%) (From Ford)

Material	Equal Stiffness	Equal Strength
Cast Iron	Base	Base
Cast Al	11	56
Cast Mg	9	64
SHEET		
Mild Steel	Base	Base
HS Steel	0	18
Aluminum	52	50
FRP	38	25
CrFRP	48	60



Successful application of reinforced polymers.

## *Fiber Characteristics, Cont'd*

Table 2: Properties of carbon, glass and Kevlar 49 fibers at 20°C

Property	Units	Carbon PAN-based Type 1	Carbon PAN-based Type II	E glass	Aromatic Polyamide Kevlar 49
Diameter	μm	7.0-9.7	7.6-8.6	8-14	11.9
Density	10 <sup>3</sup> kg m <sup>-3</sup>	1.95	1.75	0.56	1.45
Young's Modulus	GN m <sup>-2</sup>	390	250	76	125
Modulus (perpendicular to fibre axis)	GN m <sup>-2</sup>	12	20	76	
Tensile strength	GN m <sup>-2</sup>	2.2	2.7	1.4 - 2.5 (typical) 3.5 (freshly drawn)	2.8 - 3.6
Elongation to fracture	%	0.5	1.0	1.8 - 3.2 (typical)	2.2 - 2.8
Coefficient of thermal expansion (0° to 100°C)	10 <sup>-6</sup> C <sup>-1</sup>	-0.5 to -1.2 (parallel) 7-12 (radial)	-0.1 to -0.5 (parallel) 7-12 (radial)	4.9	-2 (parallel) 59 (radial)
Thermal conductivity (parallel to fibre axis)	Wm <sup>-1</sup> C <sup>-1</sup>	105	24	1.04	0.04

Notes:

1. Density of graphite single crystals is  $2.26 \times 10^3 \text{ kg m}^{-3}$ .
2. Most of the information is obtained from manufacturer's data sheets; a wide range of values has been published and this information should only be used as a rough guide.

Commercially, E-glass fibers are the preferred reinforcing fibers for polymers because of their good strength, stiffness, electrical and weathering properties, and cost.

Usually supplied in the form of continuous rovings, woven rovings, CSM, continuous strand (swirl) mat and chopped rovings.

## Matrix Materials & Their Functions in Composites

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### Thermosetting resins

e.g., unsaturated PET, phenolics, melamine, silicones, etc.

### Thermoplastics

e.g., polyamides, PC, PP, SAN, acetals, etc.

## Role of Matrix Materials in Composites?

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Stress transfer medium

Fiber surface protection, in the case of glass

Prevents brittle failure of fibers

## Voids

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Spherical or elliptical cavities parallel to the fibers.

Voids between layers of fibers and in resin-rich regions ( $0.005 \leq V_v \leq 0.02$ )

$$V_v = \frac{W_c}{r_c} - \frac{(W_c - W_f)}{r_m} - \frac{W_f}{r_f}$$

(Note:  $V_f = 1 - V_m$  iff  $V_v = 0$  or if negligible)

## Fiber Orientation

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Most important microstructural feature of FRP

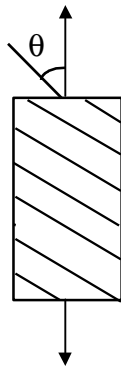
FRP are usually ANISOTROPIC



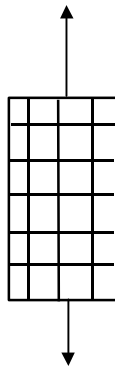
Unidirectional ( $K=1$ )



Perpendicular ( $K=0$ )



Transverse ( $K$ )



Bi-directional  
 $[K_e = 0.65$  (stiffness)]  
 $[K_\sigma = 0.30$  (strength)]

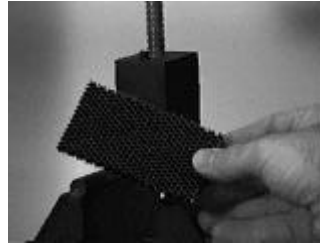
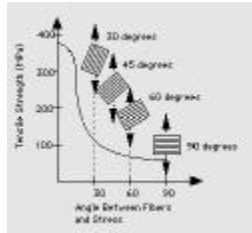
$$K = \frac{1}{a_n} \sum_{i=1}^n a_i \cos^4 q$$

= 3/8 for planar - random fibers  
 = 1/5 for random fibers in 3 - dimensions



## *Fiber Orientation, Cont'd*

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## *Planar Isotropic Composites & ROM*

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Use random planar fiber mat  
To first approximation

$$E_c = KV_f E_f + V_m E_m$$

$$\sigma_c = KV_f \sigma_f + V_m \sigma_m$$

(for perfect fiber - matrix adhesion)

Note:  $V = \phi$  in class text. Derive later.

## *Cost of Composite*

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$$m_c C_c = C_f m_f + C_m m_m + C_i m_c \text{ (\$ form } m_c)$$

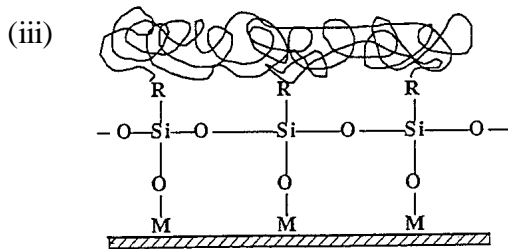
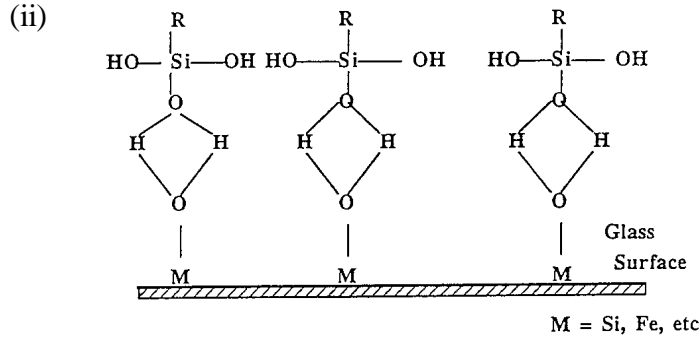
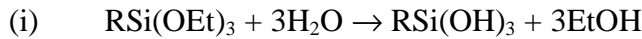
It can be shown that (see class text):

$$C_c = C_f V_f \rho_f / \rho_c + C_m (1 - V_f) \rho_m / \rho_c + C_i \text{ (\$ / unit } m_c)$$

$m$  = mass;  $C$  = cost;  $i$  = incorporation  
 $c, f, m$  = composite, fiber, matrix

## Interfacial Bonding Mechanism

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- (i) Silanol Formation
- (ii) H-Bond Formation
- (iii) Polysiloxane & Polymer-Matrix Bond Formation

Usually a monolayer (0.1 – 0.5 wt %) is used. For optimum performance of composite, 1.5 wt % of coupling agent is used because of the imperfect availability of the coupling agent due to the surface geometry and impurities.

## Role of “Sizing”

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Usually contains coupling agent or adhesion promoter {e.g.  $\text{RSi(OEt)}_3$ }  
promotes wetting between fibers and matrix  
prevents fiber-fiber contact  
protects fibers (especially glass) from corrosion during service

## Interfacial Bonding Mechanism

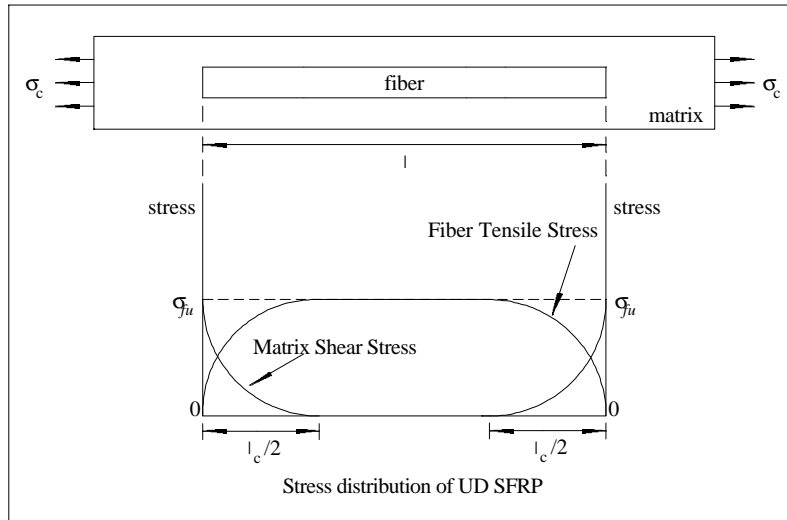
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- *Illustrate*

## *Fiber - Matrix Interface*

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Physical, mechanical or chemical bond required for efficient load bearing capacity  
Poor adhesion leads to fiber pull-out without breaking



Adhesion is less critical for E &  $\sigma$  than for toughness

As before

$$E_c = a_{\text{eff}} V_f E_f + V_m E_m$$
$$\sigma_c = a_{\text{eff}} V_f \sigma_f + V_m \sigma_m$$

( $a_{\text{eff}}$  adhesion efficiency factor)  
( $a_{\text{eff}} = 1$  for perfect adhesion)

## *End of Lecture*

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Read Class text, Ch. 6  
Optional additional reading  
>>Hull (1981)  
>>Piggott (1982)